



Investigation of the Combustion Chamber Acoustics and its Interaction with LOX/H₂-Spray Flames

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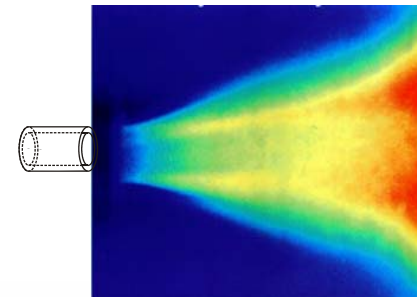
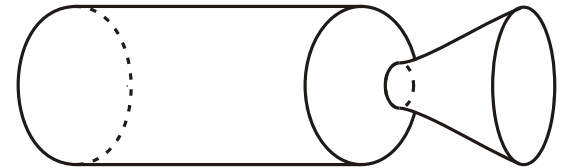
Deutsches Zentrum
für Luft- und Raumfahrt e.V.
in der Helmholtz-Gemeinschaft

HF combustion instabilities

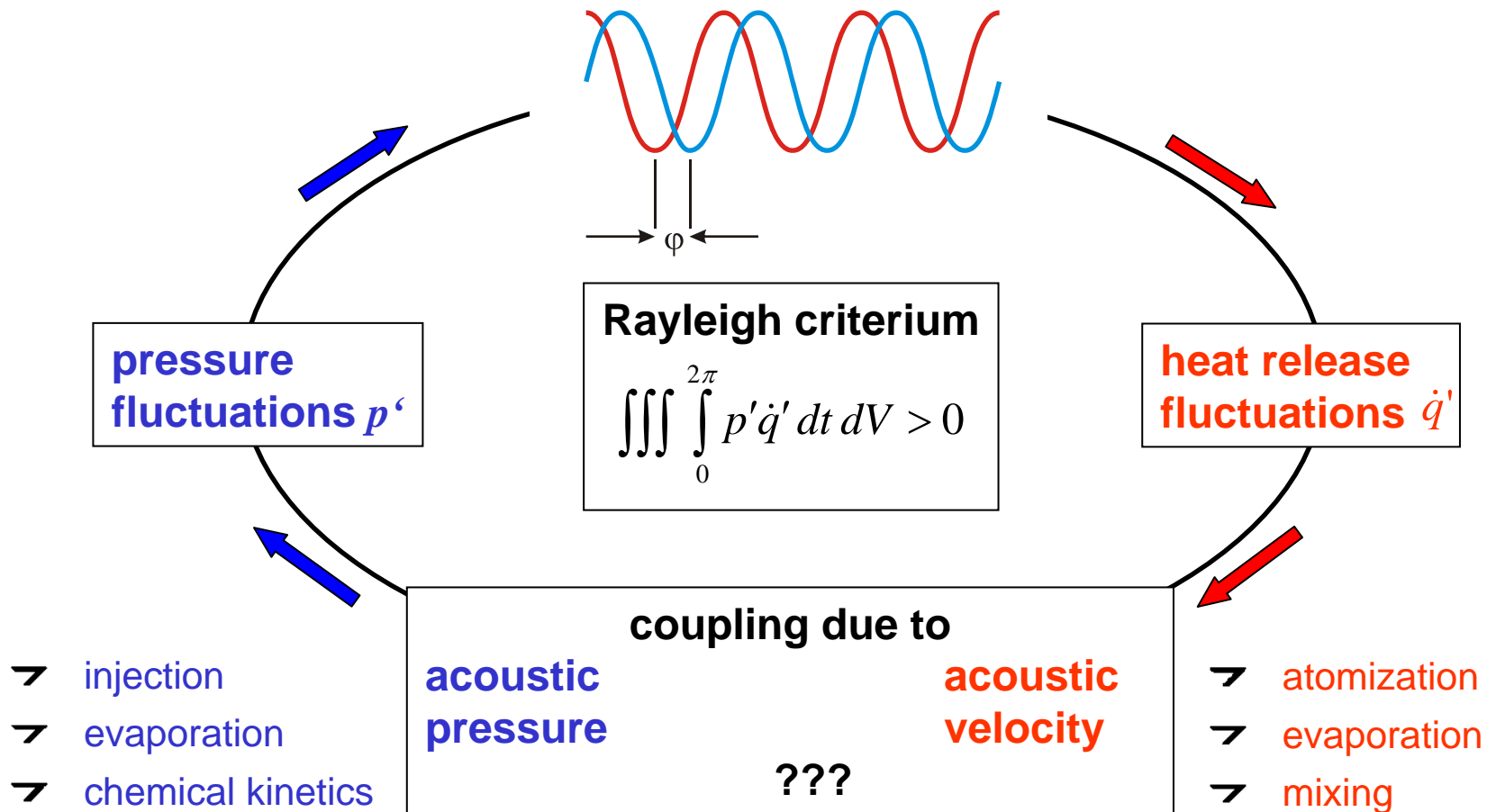
coupling of acoustic combustor resonances to combustion processes

- chamber acoustics
 - eigenmodes of a cylindrical resonator

- combustion processes
 - propellant injection
 - atomization
 - secondary atomization
 - droplet evaporation
 - chemical kinetics



coupling mechanisms



combustion chamber as acoustic resonator

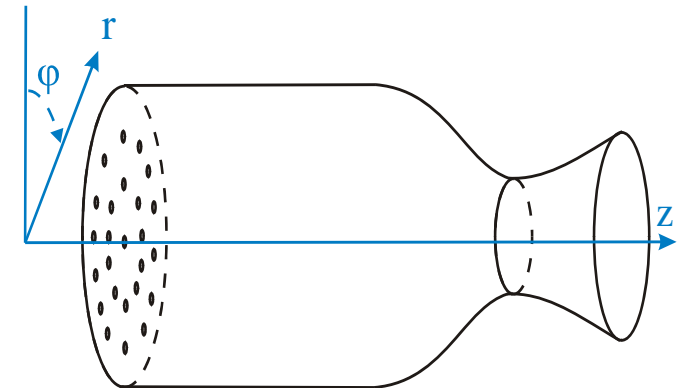
- chamber in good approximation treated as cylindrical volume
- chamber acoustically closed at the throat
- linear acoustics

$$\frac{\partial^2 P'}{\partial t^2} - c^2 \Delta P' = 0$$

- Eigen value equation

$$\Delta P' = -\frac{\omega^2}{c^2} P'$$

$$P'(z, r, \varphi) = \cos(n\varphi) J_n(\alpha_{nm} \frac{r}{R}) \cos(l\pi \frac{z}{L})$$



$l = 0, 1, 2 \dots$ longitudinal modes
 $n = 0, 1, 2 \dots$ tangential modes
 $m = 1, 2, 3 \dots$ radial modes

eigen modes of a cylindrical resonator

$$f_{nm} = \frac{\alpha_{nm} c}{2\pi R}$$

Mode	n	m	α_{nm}
1T	1	1	1.841
2T	2	1	3.054
1R	0	2	3.832
1T1R	1	2	5.331

$$R \approx 0.1m$$

➤ ambient air:

$$c = 343m/s$$

$$f_{1T} = 1.0kHz$$

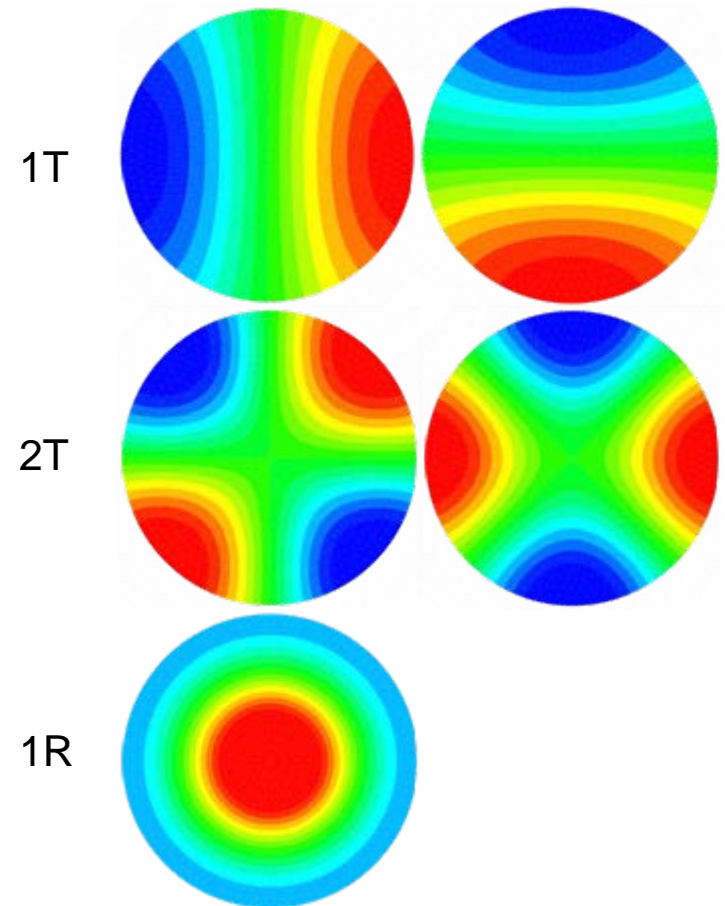
➤ hot fire test:

$$c \approx 1370m/s$$

$$f_{1T} = 4.0kHz$$

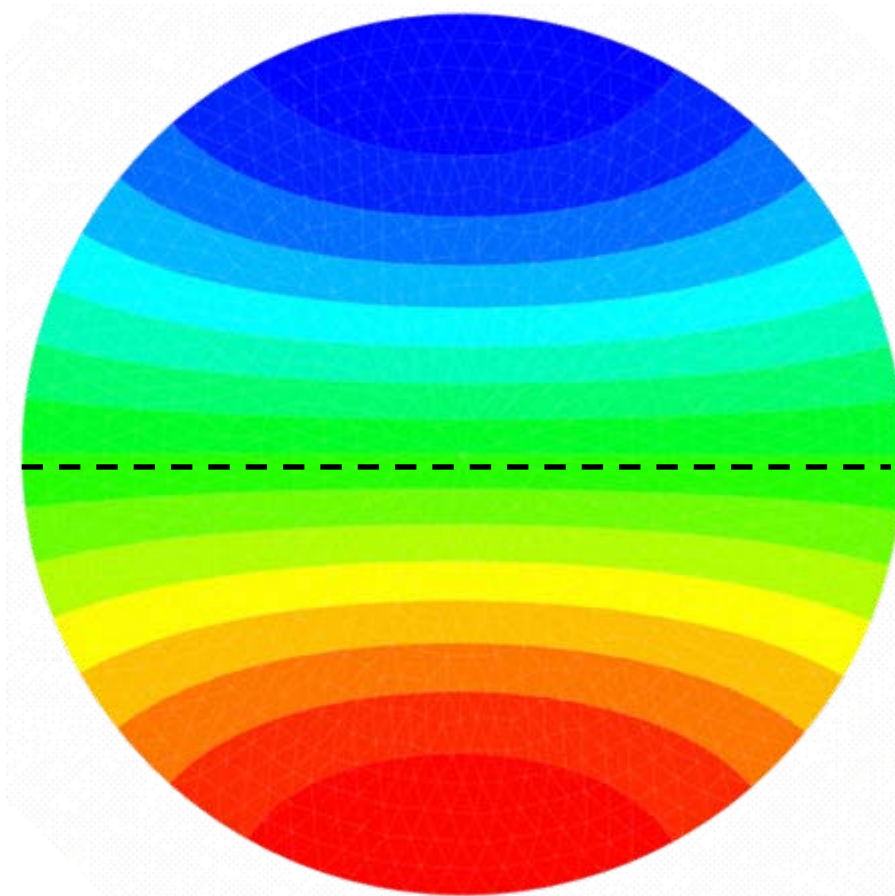
➤ tangential modes are 2-fold degenerate

➤ usually longitudinal modes not involved in HF-instabilities

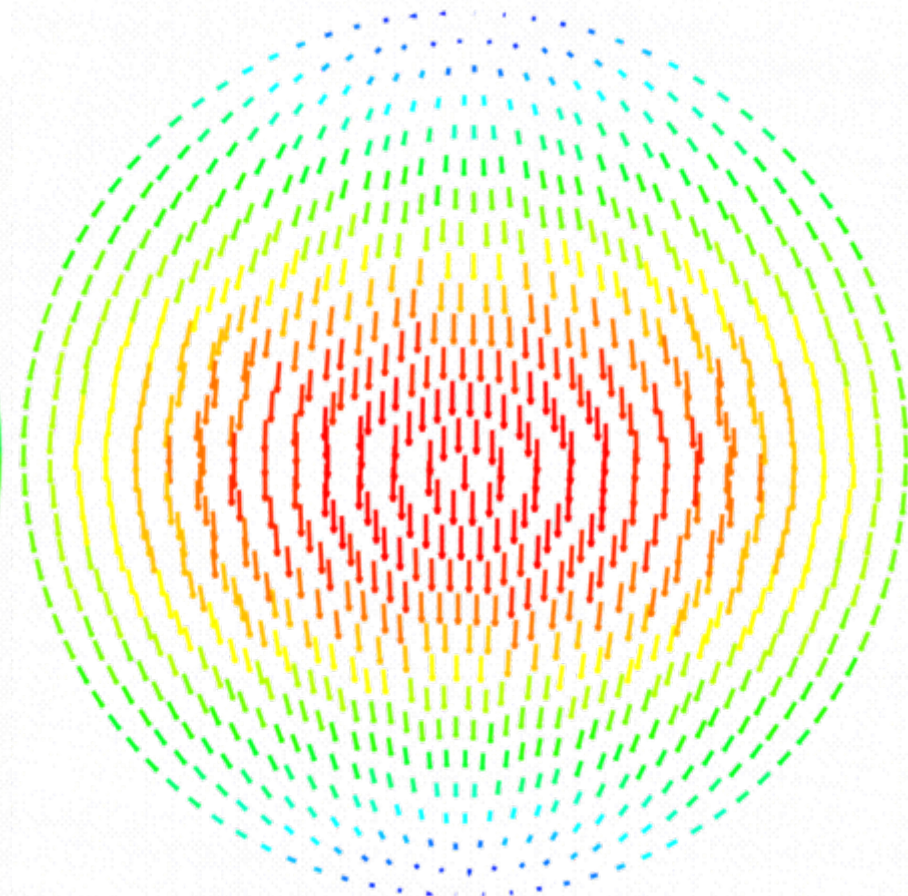


1T eigen mode

pressure field

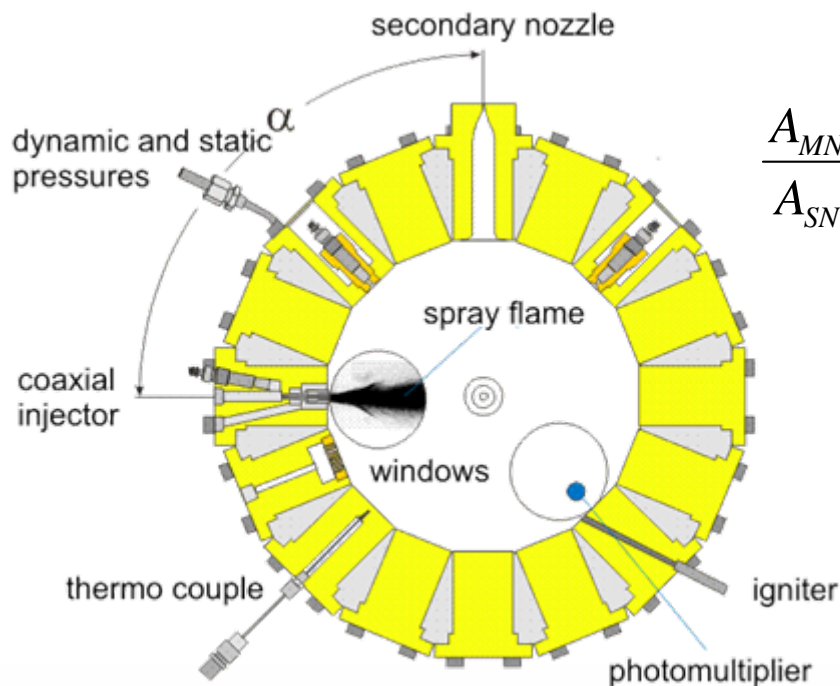
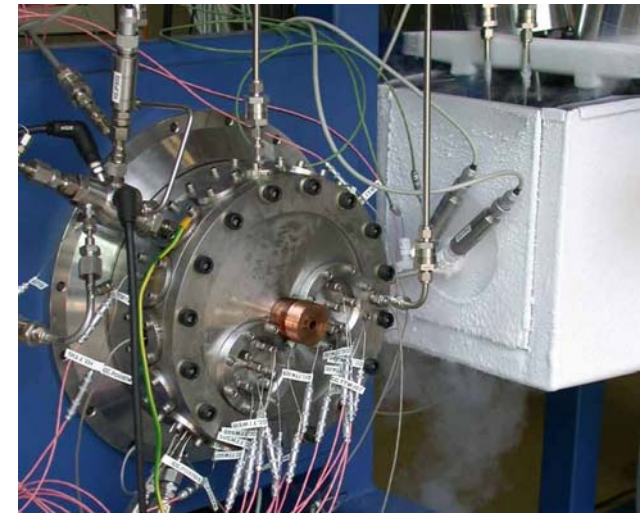


velocity field

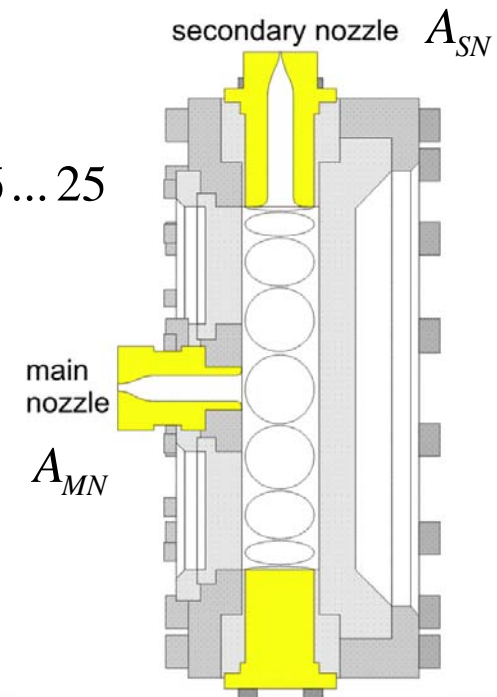


HF-combustor (CRC)

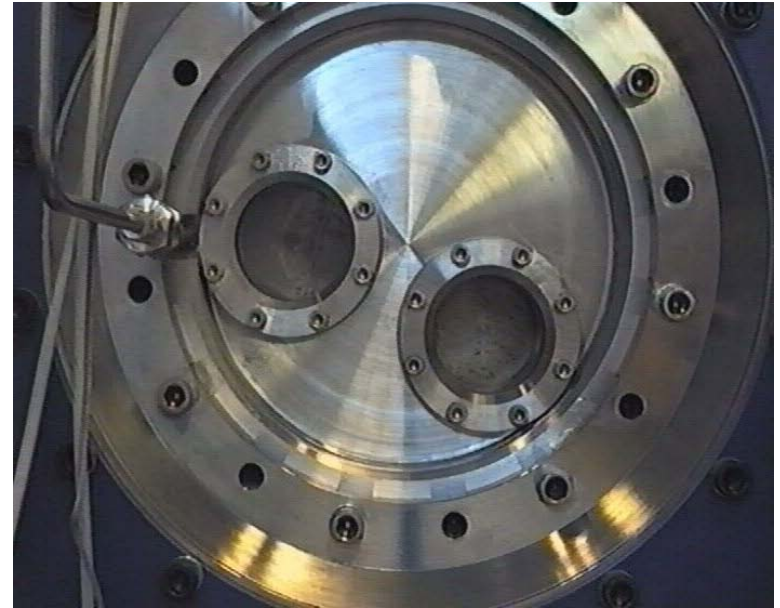
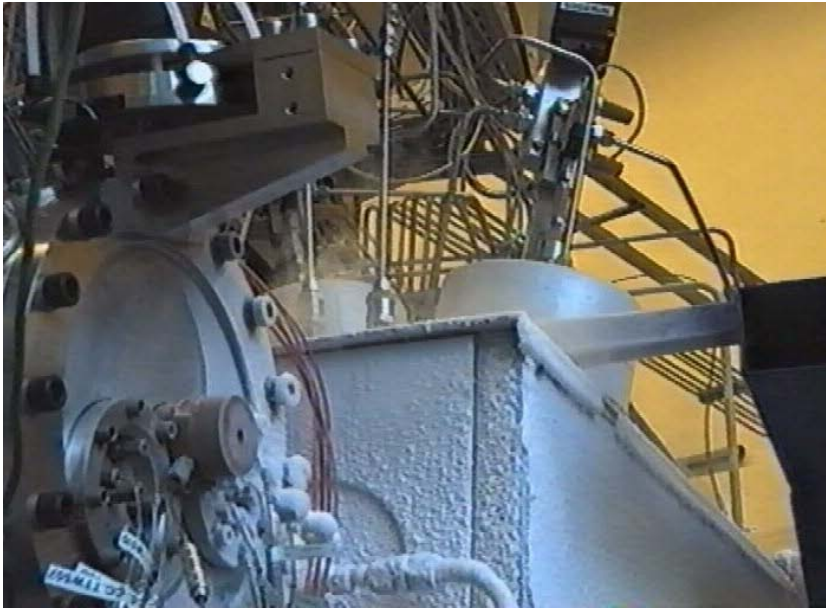
- diameter x height: 20 cm x 4 cm
- transversal modes at representative frequencies
- p_c up to 10 bar
- LOX, GH_2 @ 77K
- coaxial injector
- injection in radial direction



$$\frac{A_{MN}}{A_{SN}} = 6 \dots 25$$

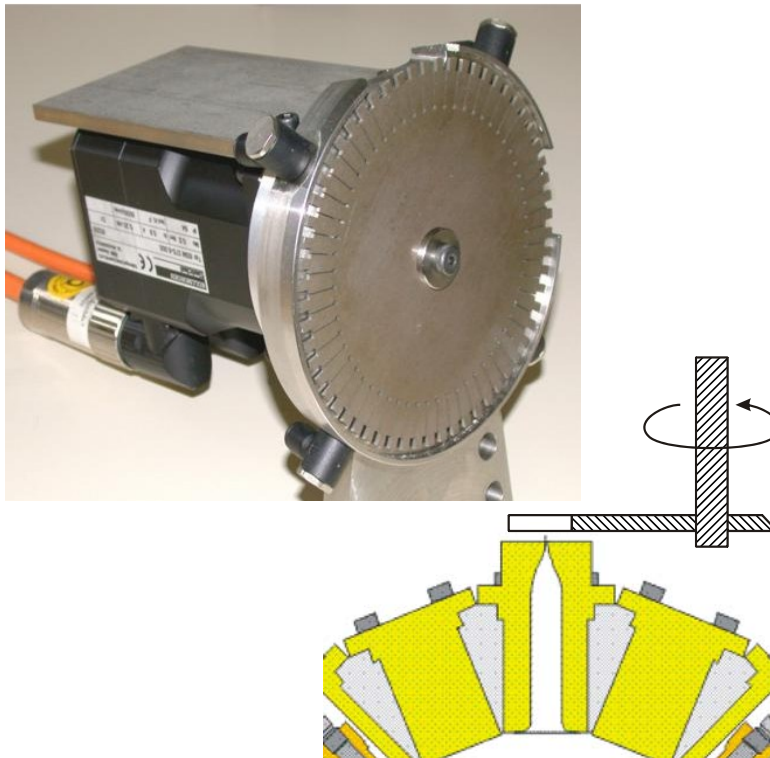


LOX/H₂-spray combustion in the CRC

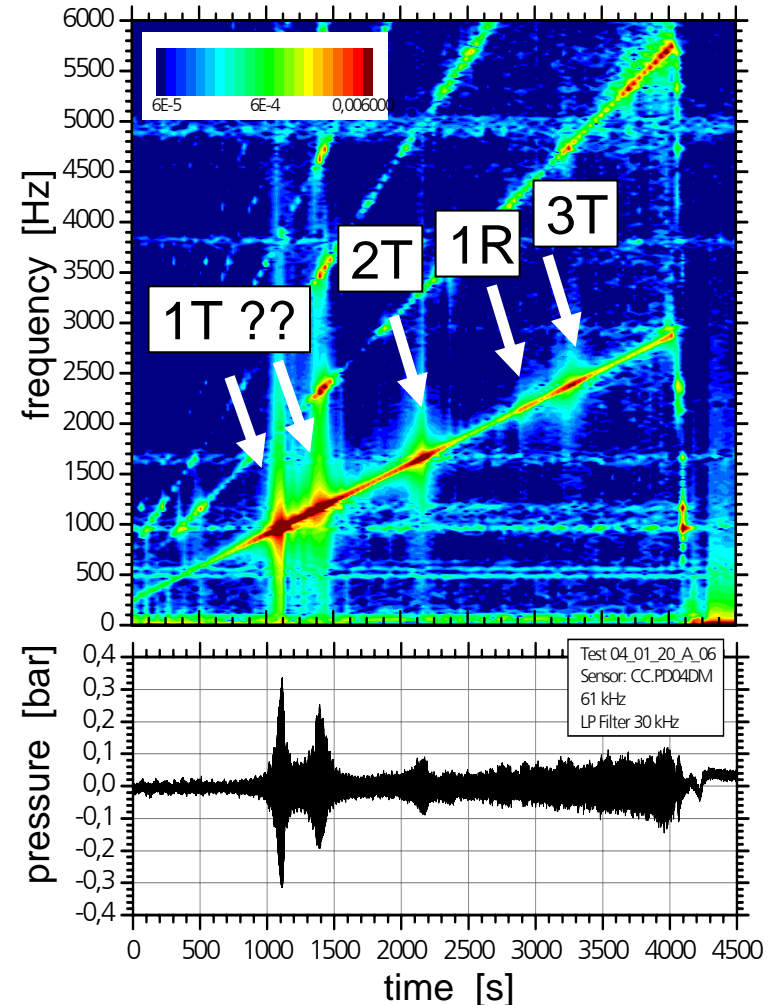


external excitation by siren ambient air

- siren at secondary nozzle exit
- frequency ramping during test with 500 Hz/s

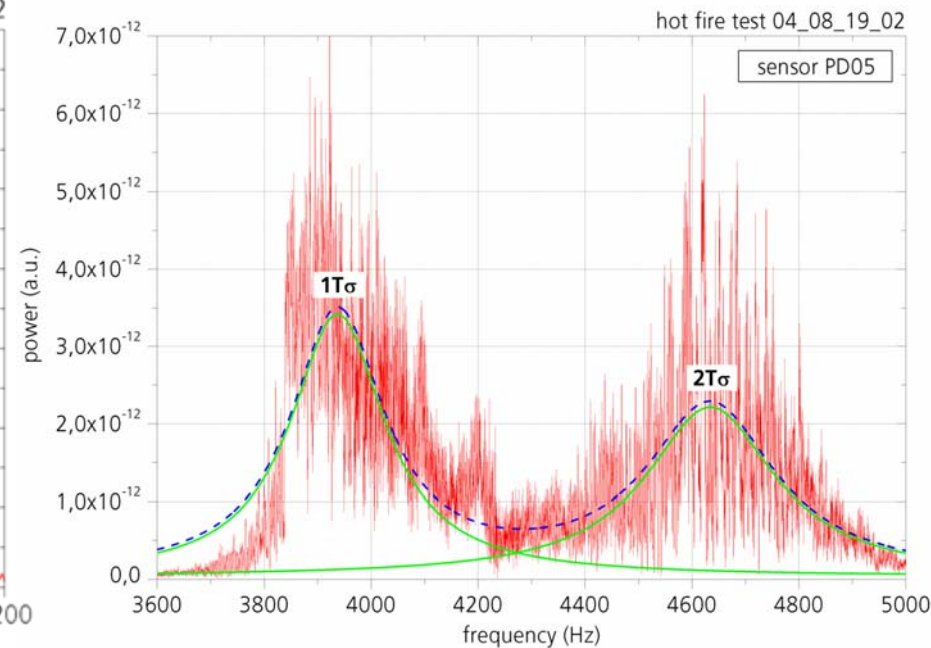
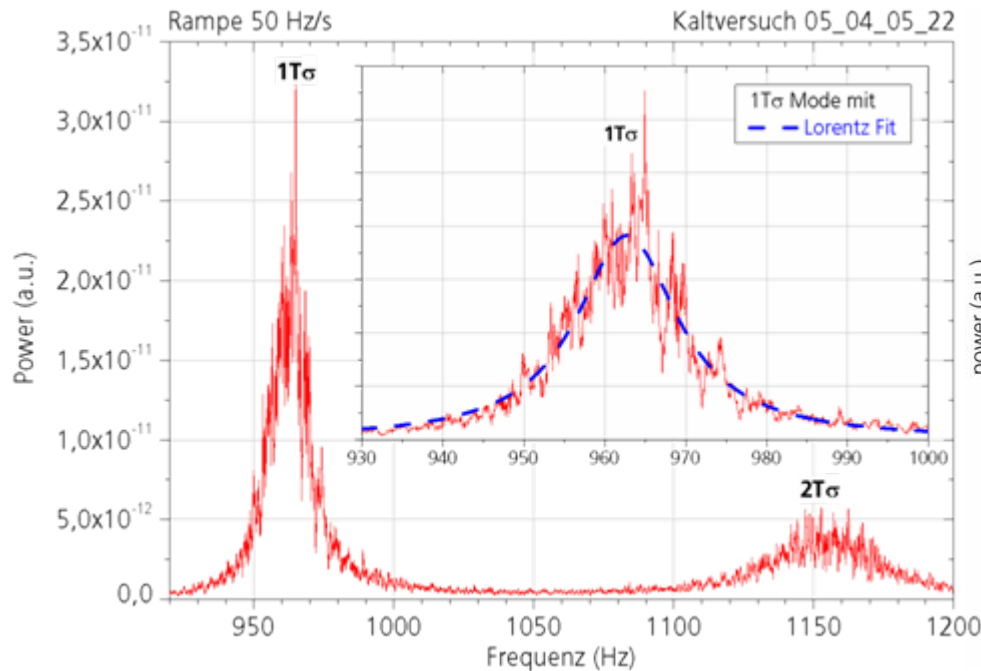


mode	(349 m/s)
1T	1022 Hz
2T	1695 Hz
1R	2127 Hz
3T	2332 Hz



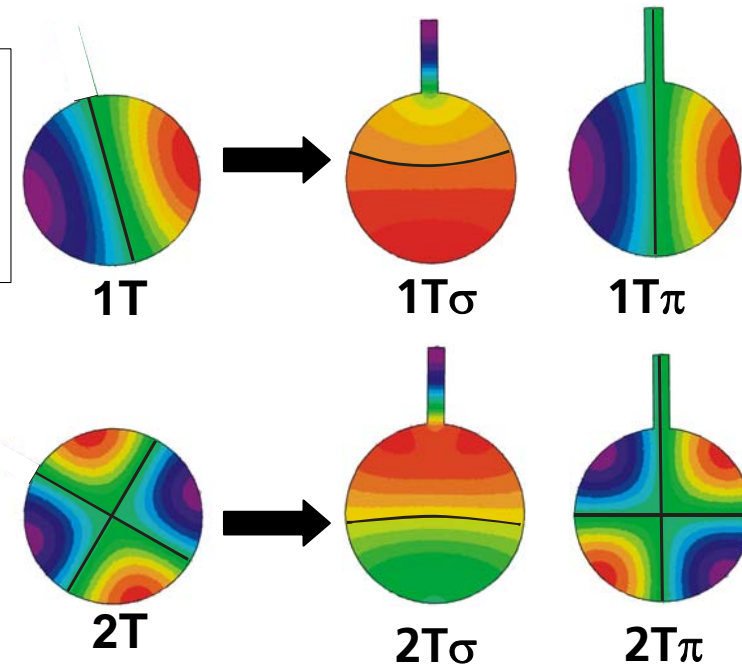
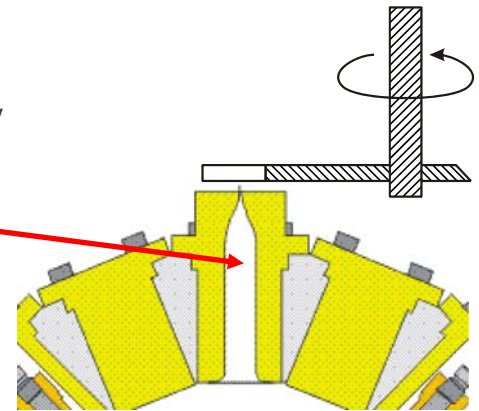
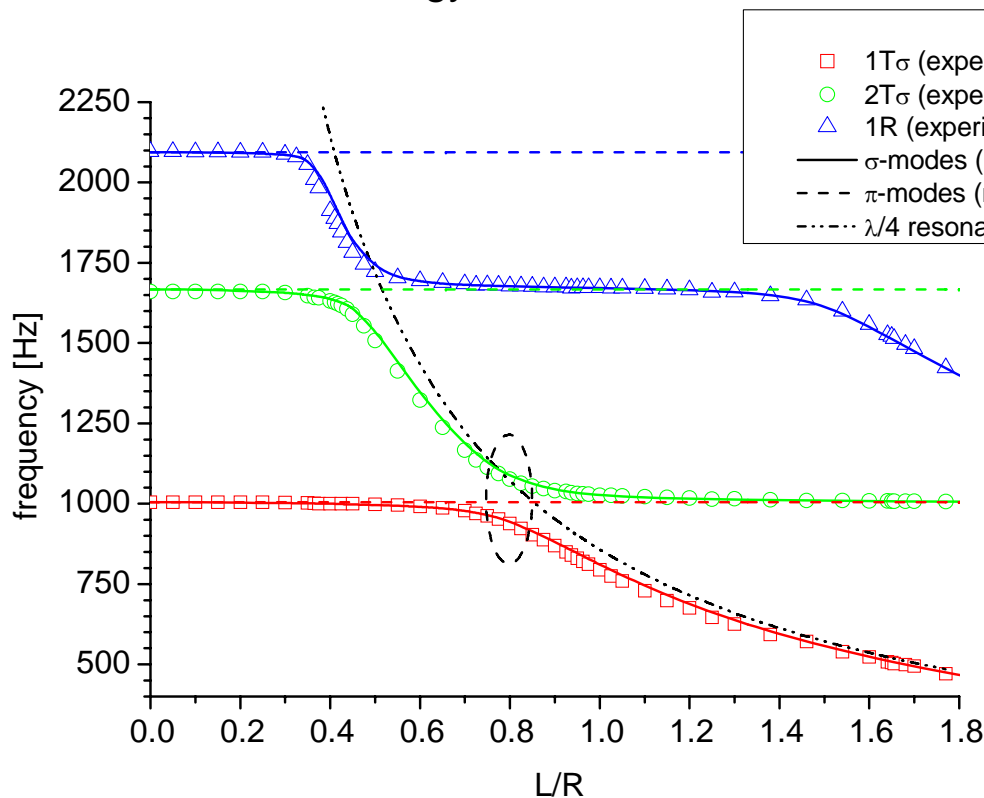
resonances near 1T-cylinder mode in combustor with secondary nozzle

- "noisy" appearance of resonances with siren excitation
- hit fire tests: line width and asymmetric profile indication for combustion/acoustics interaction



eigenfrequencies of cylinder with cavity

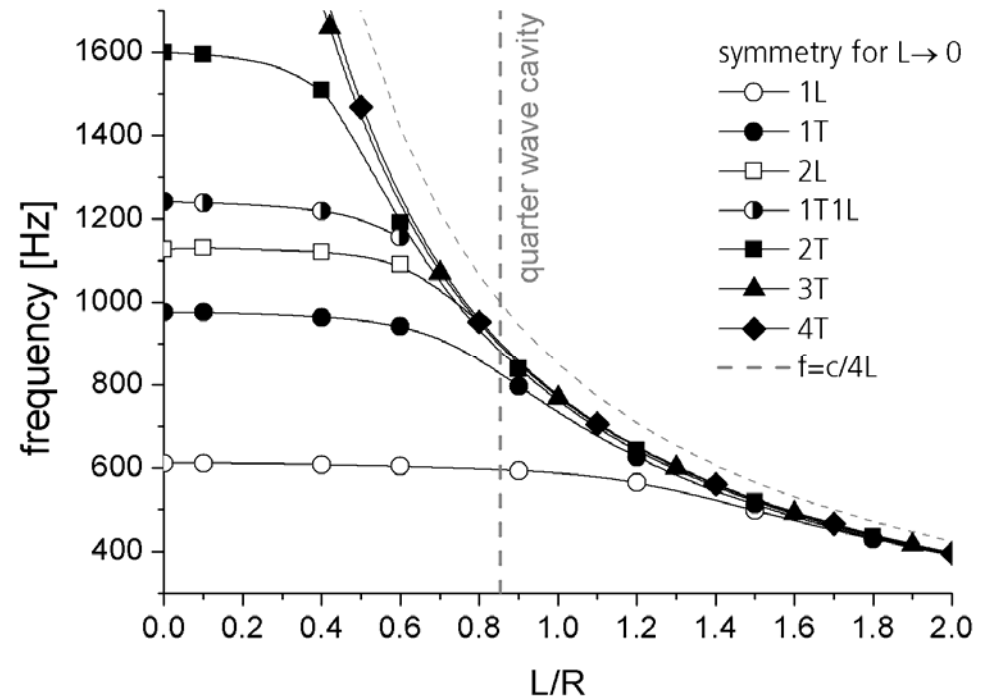
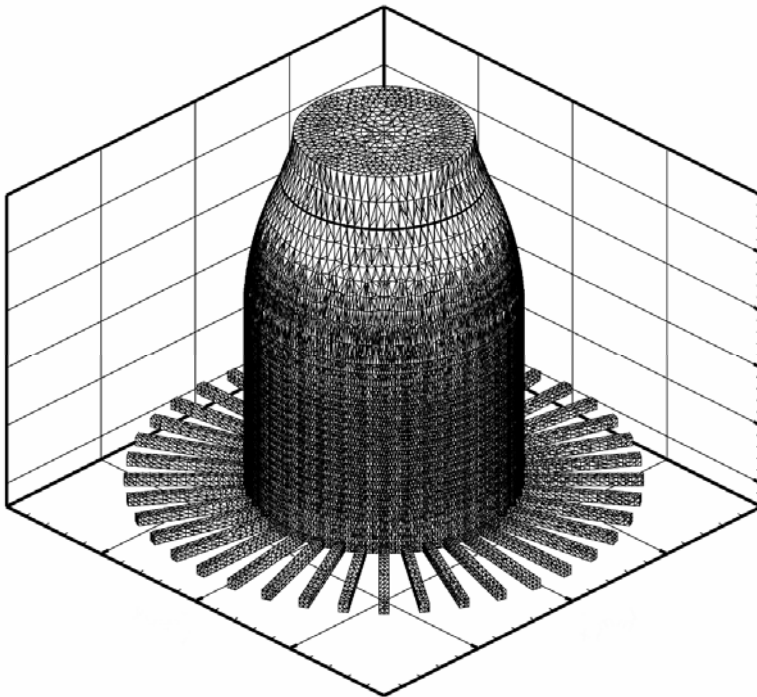
- secondary nozzle forms additional resonance volume
- cavity breaks 2-fold degeneracy of tangential eigenmodes
 - two eigenmodes of different symmetries and eigenfrequencies
- N.B.: analogy to $\lambda/4$ -absorbers



M. Oswald, Z. Farago, G. Searby, F. Cheuret,
 "Resonance frequencies and damping of a cylindrical
 combustor acoustically coupled to an absorber",
 submitted to Journal of Proulsion and Power, June 2007

eigenfrequencies for combustor with absorber ring

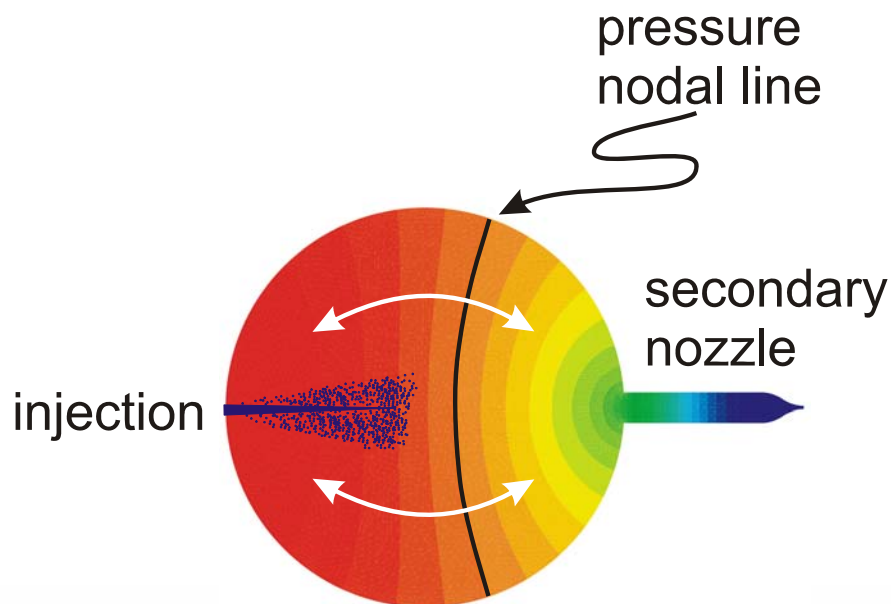
- first 8 low lying modes analyzed
- absorber ring detunes acoustic system
- modes group near to the cavities $\lambda/4$ -resonance



pressure distribution for SN at 90° and 180°

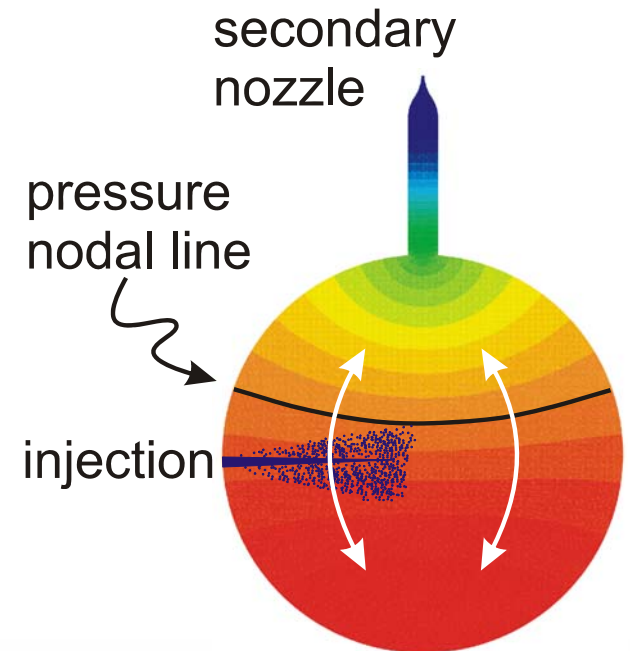
SN at 90°

- spray exposed to strong acoustic pressure fluctuation
- no transversal velocity fluctuation

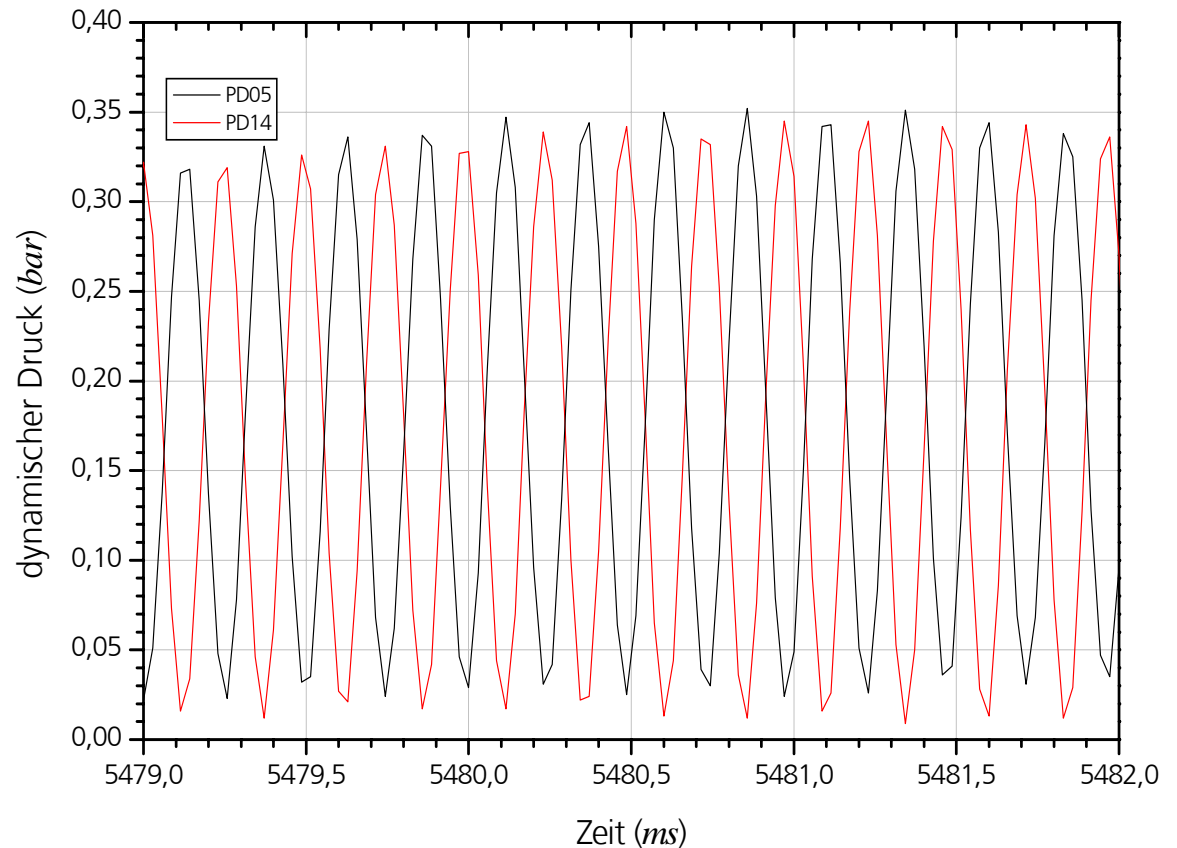
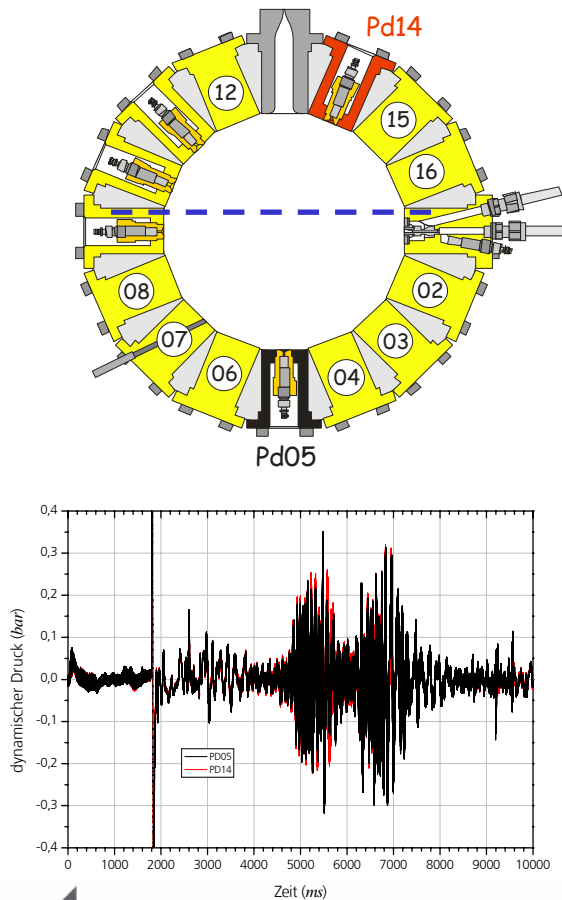


SN at 180°

- spray exposed to strong transversal acoustic velocity fluctuations
- very small pressure fluctuations

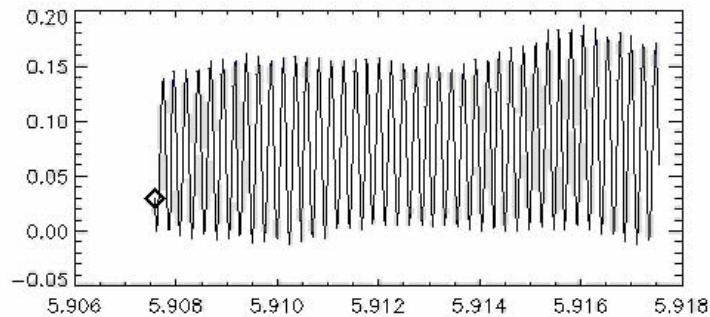
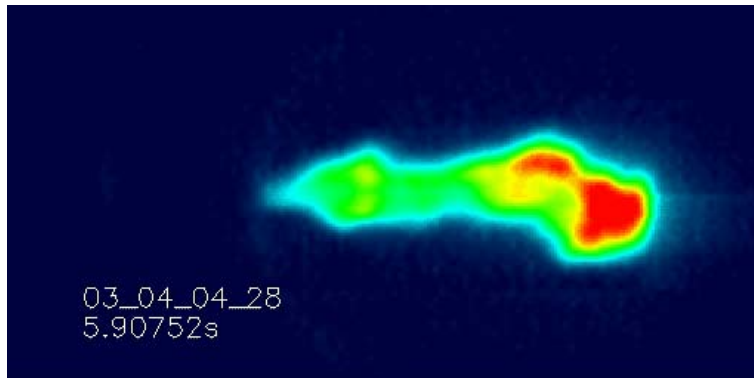


dynamic pressure on resonance for SN at 90°

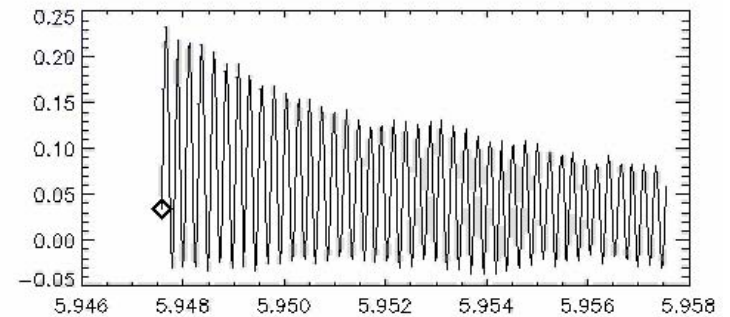
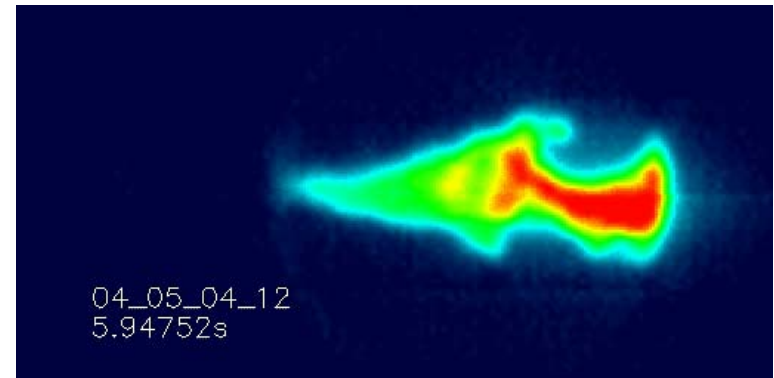


OH-chemiluminescence during excitation

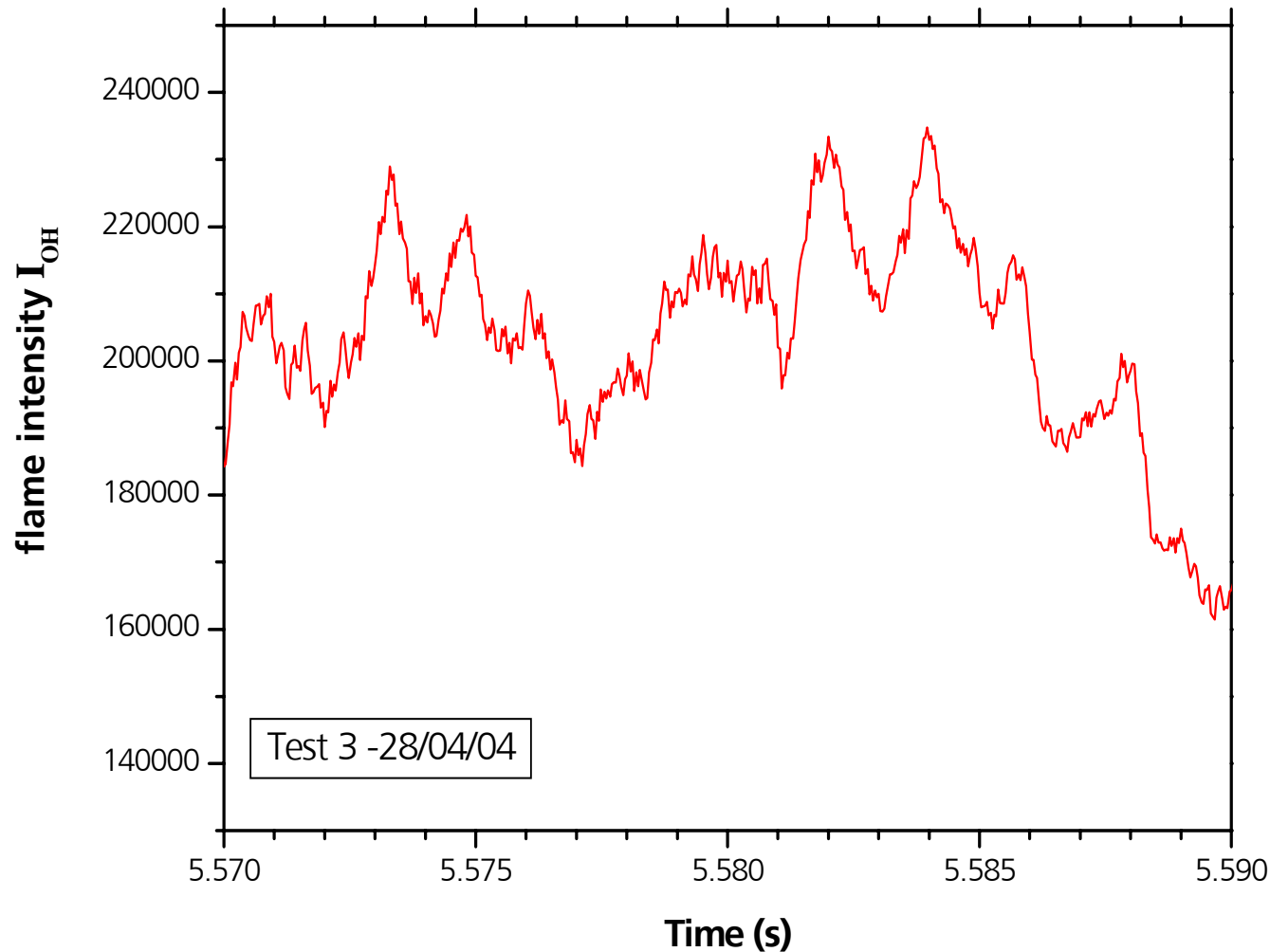
excitation with SN at 90°



excitation with SN at 180°



OH-chemiluminescence during excitation

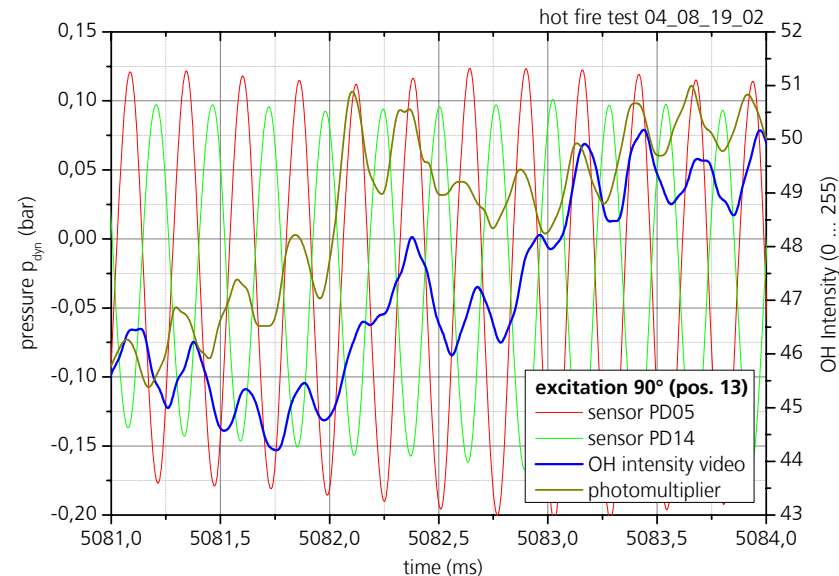
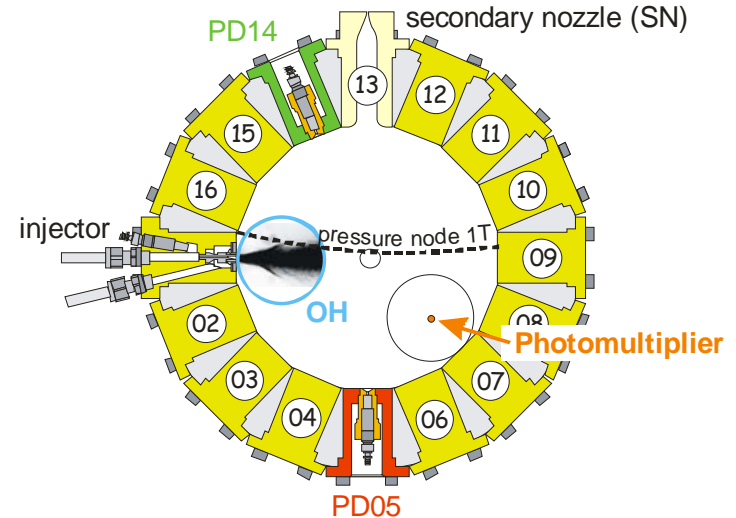


intensified CCD, 27KHz, filter with transmission at 300-310nm



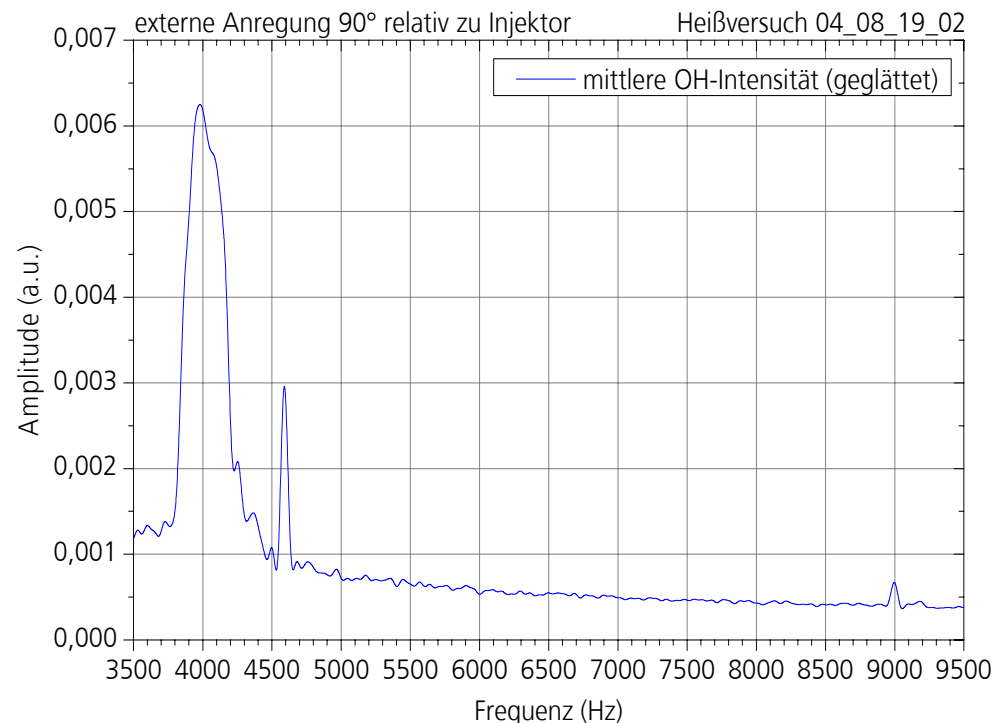
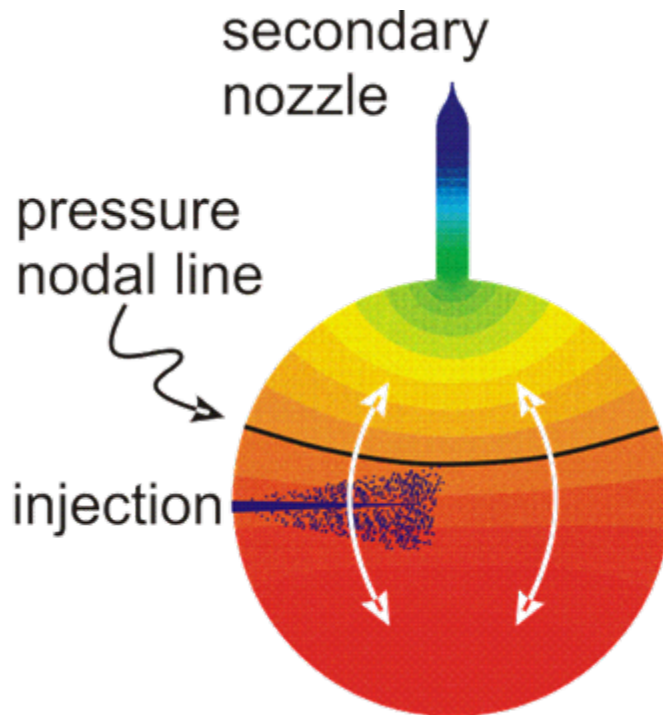
combustion response

- fluctuation of pressure and OH-emission in phase:
 - positive coupling of combustion and acoustics (Rayleigh criterium)
- excitation levels up to $p'/p \sim 20\%$: no combustion instability observed
 - damping effects suppress gain due to positive coupling
- excitation level did not show any correlation with injection conditions (J , We , R_v , ...)
 - no coupling of acoustics to spray processes in the experiments



combustion response

- at excitation with SN at 90° no significant frequency component at 2ω
 - coupling due to pressure sensitive processes



interaction index

➤ Rayleigh Kriterium: $\int_V \int_t Q' \cdot P' dt dV$

➤ response factor:

$$p = \bar{p} + p' \cdot \cos(\omega t)$$

harmonic time dependency

$$\bar{Q} = \bar{\dot{Q}} + \dot{Q}' \cdot \cos(\omega t + \varphi)$$

$$N = \frac{\int_V \int_t (Q' / \bar{Q}) \cdot (p' / \bar{p}) dt dV}{\int_V \int_t (p' / \bar{p})^2 dt dV} = \frac{\dot{Q}' / \bar{\dot{Q}}}{p' / \bar{p}} \cdot \cos(\varphi)$$

$$N = \frac{I' / \bar{I}}{p' / \bar{p}}$$

chemiluminescence = heat release

$\varphi=0$

mean flame intensity for SN at 90°

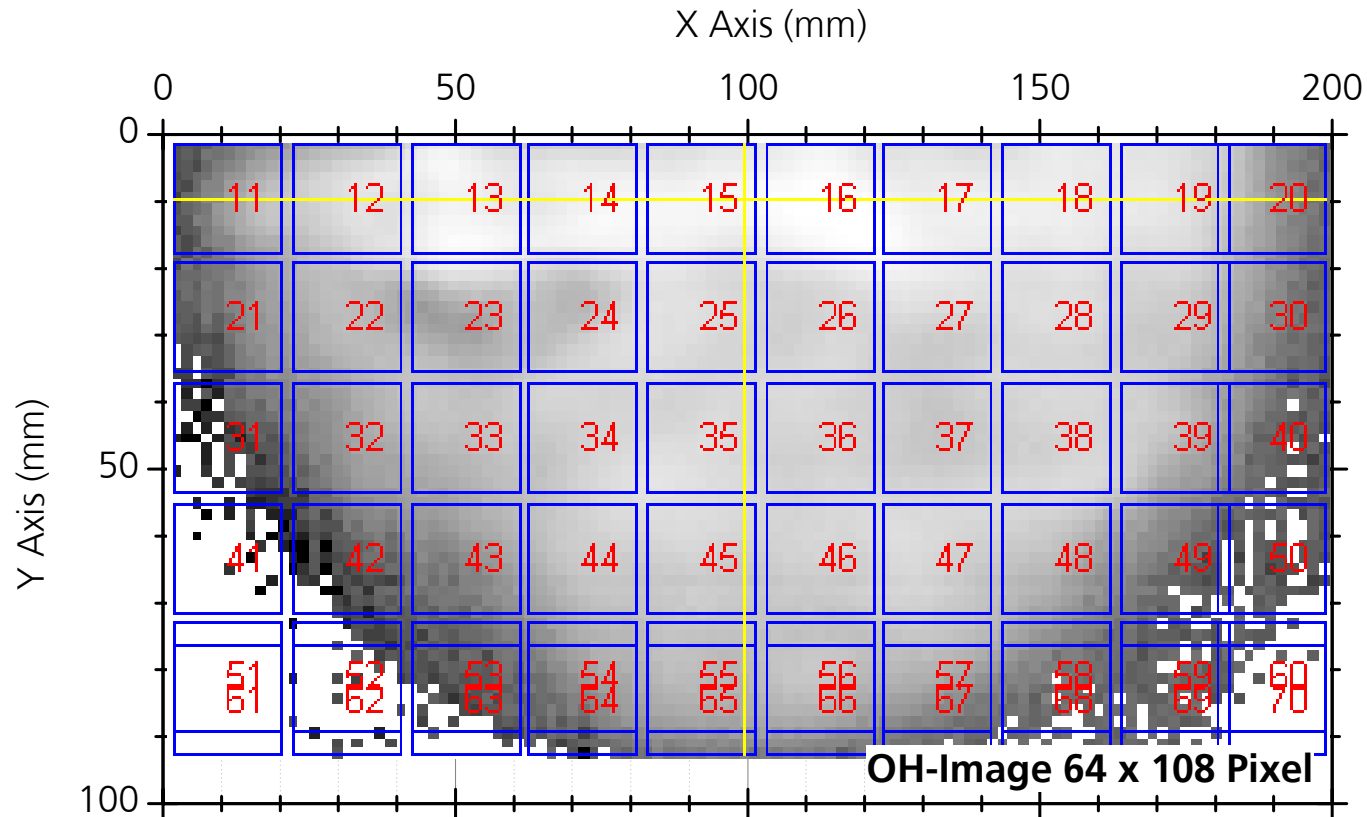
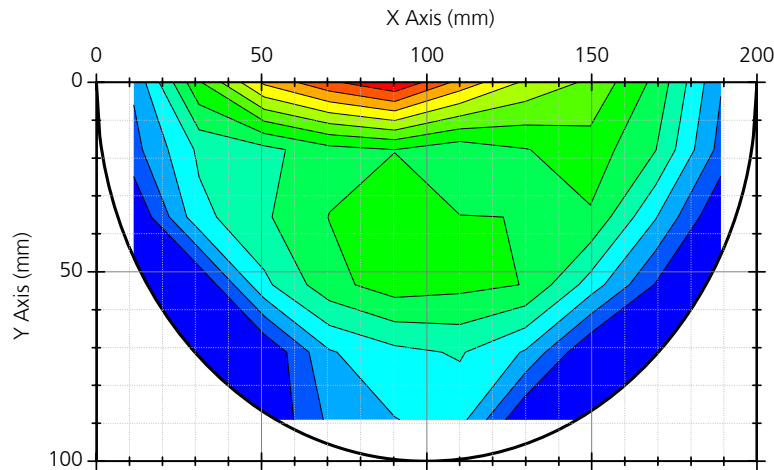


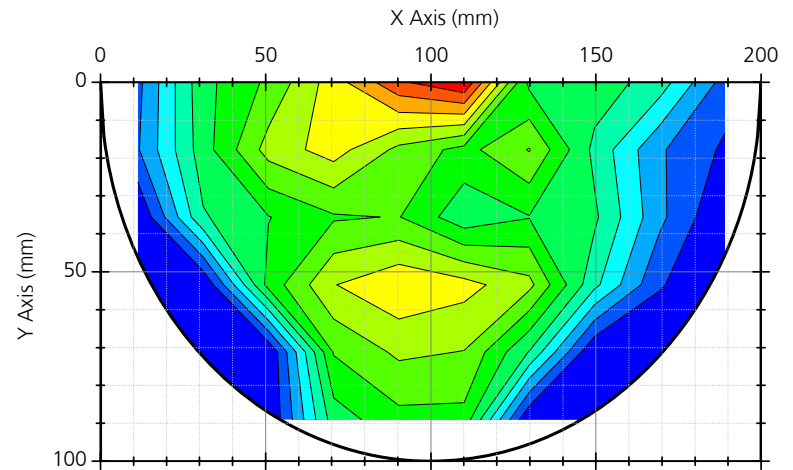
image processing on sub areas

- mean intensity I
- fluctuation I'
- reconstruction of pressure field p, p' from dynamic pressure sensors

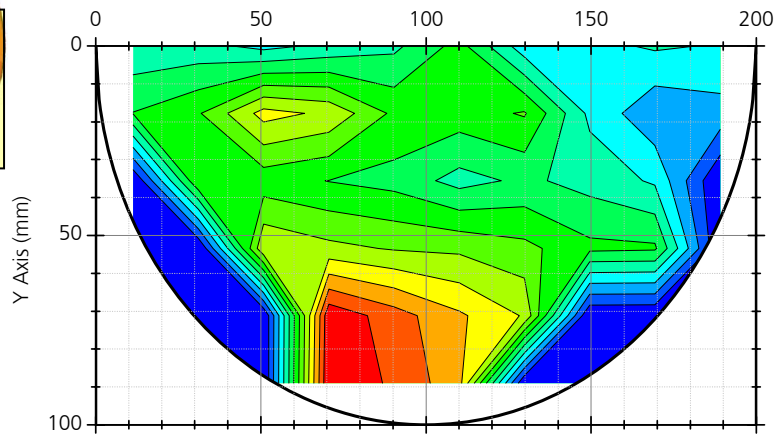
interaction index for SN at 90°



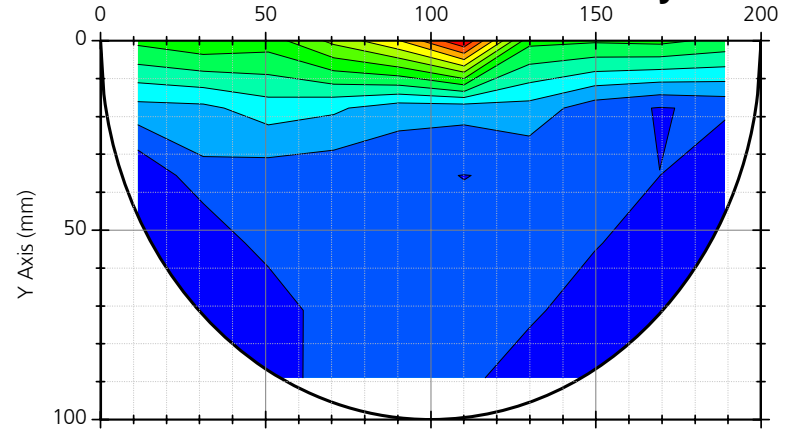
mean flame emission I



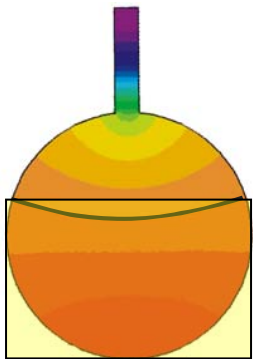
fluctuation of flame intensity I'



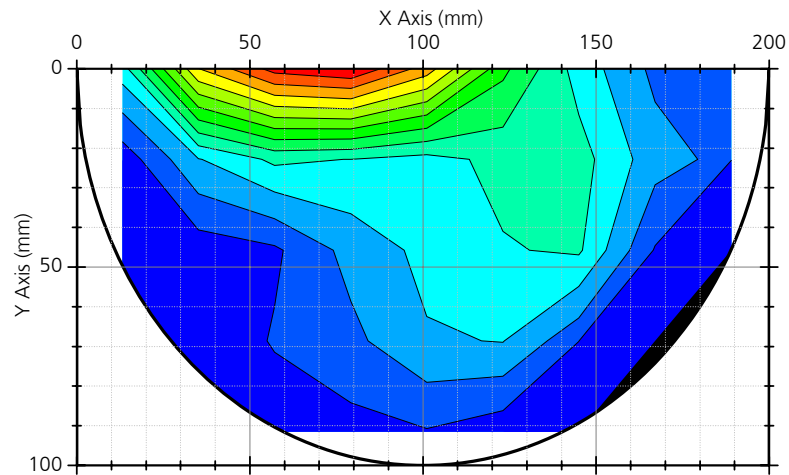
relative flame response I'/I



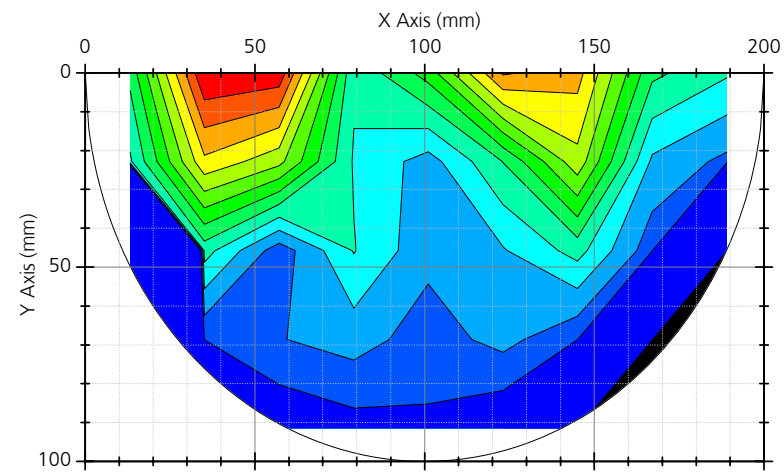
interaction index $(I'/I)/(p'/p)$



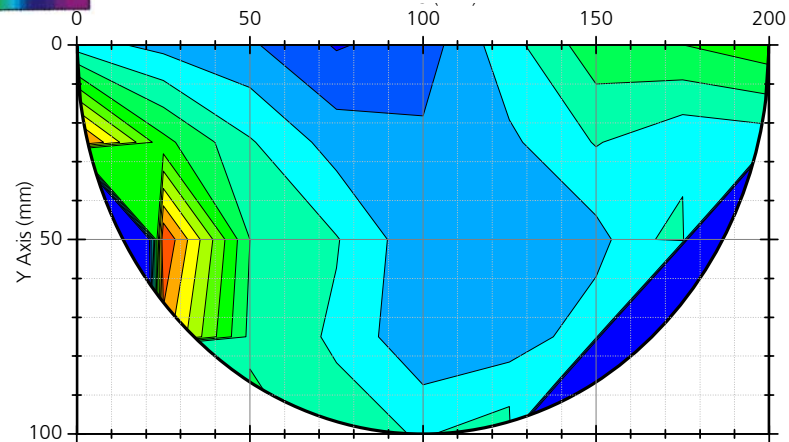
interaction index for SN at 180°



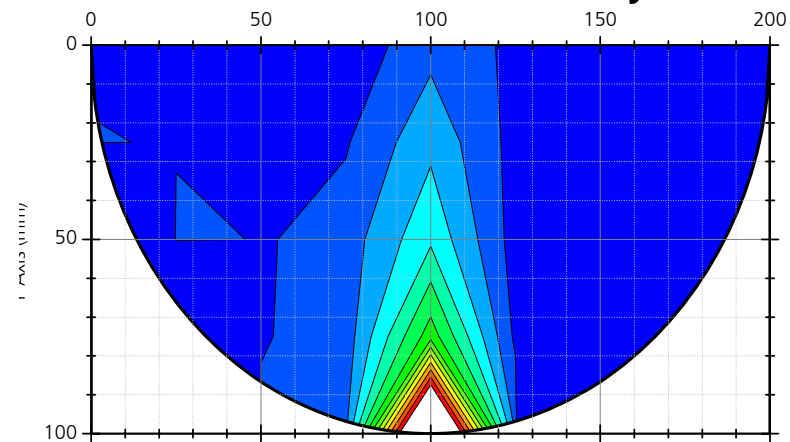
mean flame emission I



fluctuation of flame intensity I'



relative flame response I'/I



interaction index $(I'/I)/(p'/p)$